Fatigue Crack Growth on Several Materials under Single-Spike Overloads and Aircraft Spectra

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International Conference on Aeronautical Fatigue 26-29 June 2023 Delft, The Netherlands





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• Plane-strain to plane-stress fatigue-crack growth behavior

Flat-to-Slant Crack Growth and the Associated Constraint-Loss Behavior

ASTM STP-415: Crack-growth rate was "constant" at transition on 2024-T3 Alclad for wide range in R

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- Fatigue-crack-growth-rate against ΔK_{eff} correlations

FASTRAN – Fatigue-Crack-Closure based Life-Prediction Code 1976 - Present



Effective SIF Relation for 9310 Steel Plate C(T) Specimens



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- Single-spike overload/underload tests and analyses
 - 9310 Steel Plate C(T) Newman et al. (2013)

Repeated Single-Spike Overload/Underload History under Constant-Amplitude Loading



Measured and Predicted Crack-Length-against-Cycles for C(T) Specimen made of 9310 Steel Plate under Repeated Single-Spike Overloads



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 - 2024-T3 Sheet M(T) Newman-Walker

Predicted Crack-Length against Cycles under Repeated Single-Spike Overloads in 2024-T3 Sheet



Measured and Predicted Crack-Length against Cycles under Repeated Single-Spike Overloads in 2024-T3 Sheet



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 - 2024-T3 Sheet M(T) Newman-Walker
 - 7075-T6 Sheet M(T) Newman-Walker

Measured and Predicted Crack-Length against Cycles under Repeated Single-Spike Overloads in 7075-T6 Sheet



Measured and Calculated Constraint-Loss Behavior for 7075-T6 Sheet under Constant-Amplitude Loading



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 - 2024-T3 Sheet M(T) Newman-Walker
 - 7075-T6 Sheet M(T) Newman-Walker

TWIST spectrum crack-growth tests (Wanhill) and analyses

Crack Growth under TWIST (Level III) Spectrum Loading

Wanhill (1977), Newman (1992)



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- Constraint-loss behavior from plane-strain to plane-stress behavior was able to calculate or predict delays caused by single-spike overloads and underloads.
- **Constraint-loss behavior** from plane-strain to plane-stress behavior was *able* to predict crack growth under the **TWIST** spectrum loading.

Thank You, Very Much !

Questions ?





